

Abstract

2.5D SiO_{2f}/SiO₂ is extensively utilized in the aerospace industry. The performance under high-temperature cyclic fatigue is paramount importance for potential applications. This study examines the fatigue behavior at service temperatures, alongside the residual strength. Findings reveal that temperature elevation enhances both the tensile strength and fatigue resistance. Under reduced fatigue loads, specimens exhibited signs of fatigue strengthening, demonstrating augmented strength and stiffness compared to their initial state. However, the interfacial bond strength in these specimens decreased, leading to a shift in fracture behavior from purely brittle to a combined fracture mode.

Introduction

Quartz fiber reinforced silica composite (SiO_{2f}/SiO₂) is widely used in antenna covers in the aerospace field [1]. During flight, the antenna cover is subjected to the combined effects of high temperature, high pressure, vibration. It is necessary to comprehensively analyze the influence of thermodynamic coupling load on the material's performance [2]. Scholars have conducted a series of studies on the thermal and mechanical properties of 2.5D SiO_{2f}/SiO₂ [3], [4]. The bending performance [5], shear performance [6], and ablation performance [7] at high temperatures have been studied, which reveal that high temperature can form a glassy SiO₂ film on the material surface, affecting the fracture mode [8]. In summary, the physical and mechanical properties of 2.5D SiO_{2f}/SiO₂ have strong temperature dependence. However, there have been few reports on the high temperature fatigue performance.

In this study, fatigue performance tests were carried out on 2.5D SiO_{2f}/SiO₂ at elevated temperatures. The influence of temperature and fatigue load on fatigue behavior, and the influence of fatigue load on residual strength were investigated. A residual stiffness degradation model was used. The damage mechanism of the material under high temperature fatigue was discussed in combination with fracture morphology.

Material and Method

Fatigue temperature: 800°C, 900°C, 1000°C
Frequency: 5Hz
Upper limit: 10⁵ cycles
Scanning electron microscope: Navo Nano SEM-450
Fatigue test and tensile strength equipment: high-temperature furnace: HTF06-1400
Stress ratio: 0.1
Loading rate of static strength test: 0.5mm/min
3D X-ray microscope: Zeiss Xradia 510 Versa
SDS-100 electro-hydraulic servo fatigue testing machine

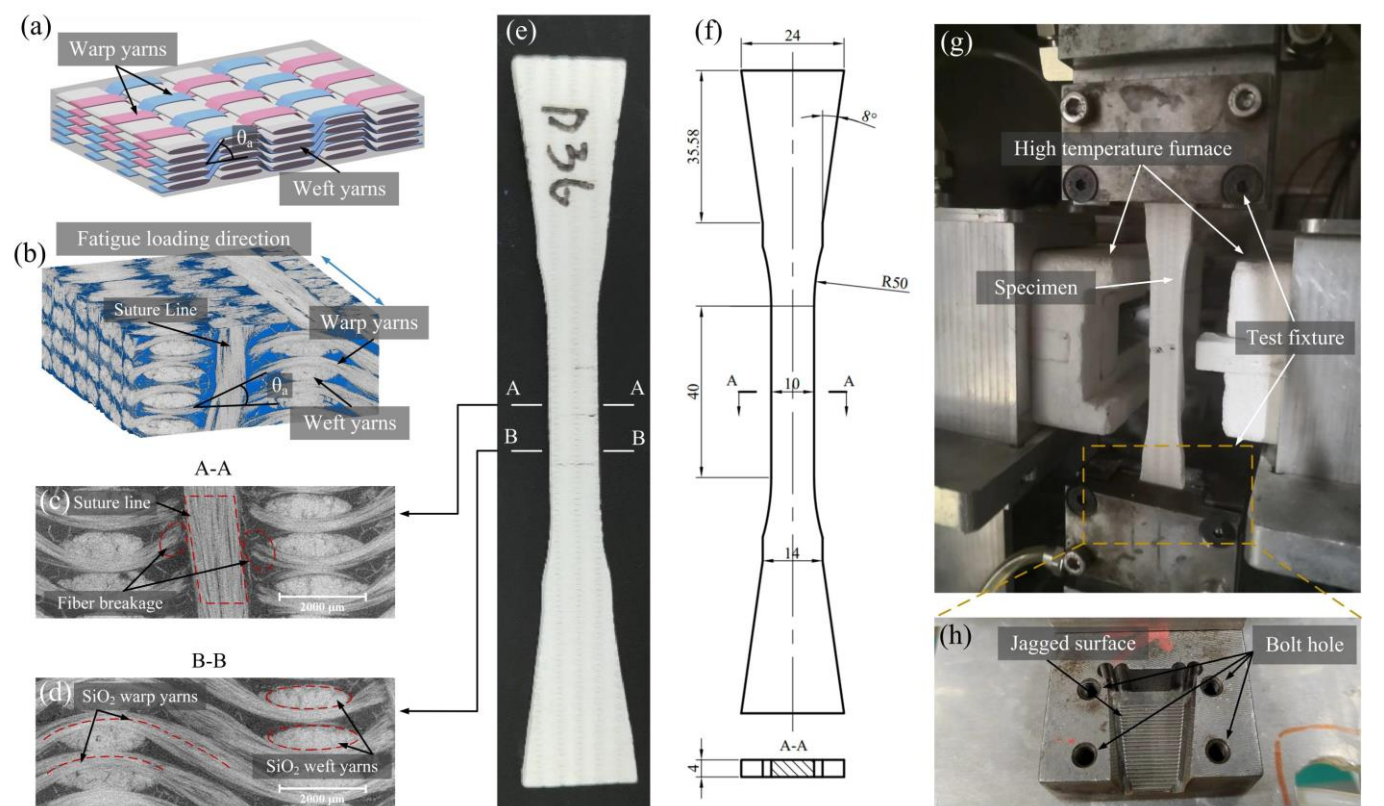


Figure 1. Microstructure and specimen diagram of 2.5D SiO_{2f}/SiO₂. (a) Schematic of fiber preform. (b) CT reconstruction diagram. (c) Longitudinal section of the specimen (with sutural line). (d) Longitudinal section of the specimen (without sutural line). (e) Photo of the specimen. (f) Size of the specimen. (g) Fatigue test equipment. (h) Test fixture.

Results

1. Effect of temperature on mechanical properties of materials

Considering the service temperature of the radome, along with the long-term service temperature of SiO_{2f}/SiO₂ being 1000°C, this study selected the range of 800°C to 1000°C for the investigation of fatigue behavior and residual strength. A set of six static tensile strength tests were performed, listed in Table 1.

Table 1. Tensile strength at different temperatures

T/°C	RT	800	1000
Average strength	24.46	30.87	34.23

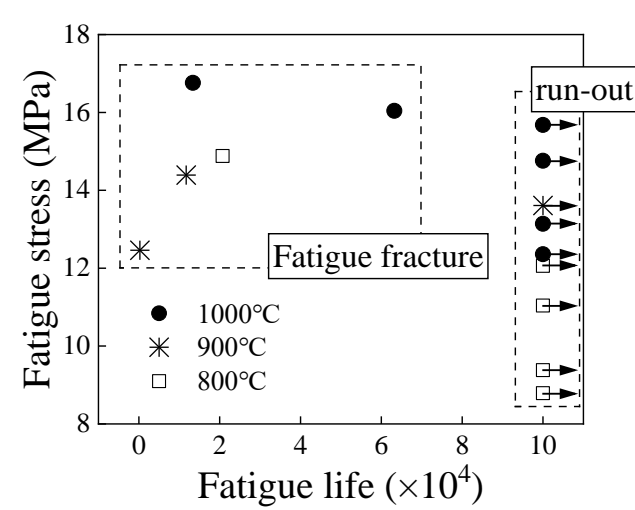


Figure 2. Fatigue life of material at different temperatures and stress levels

2. Typical hysteresis loops and residual stiffness at elevated temperature

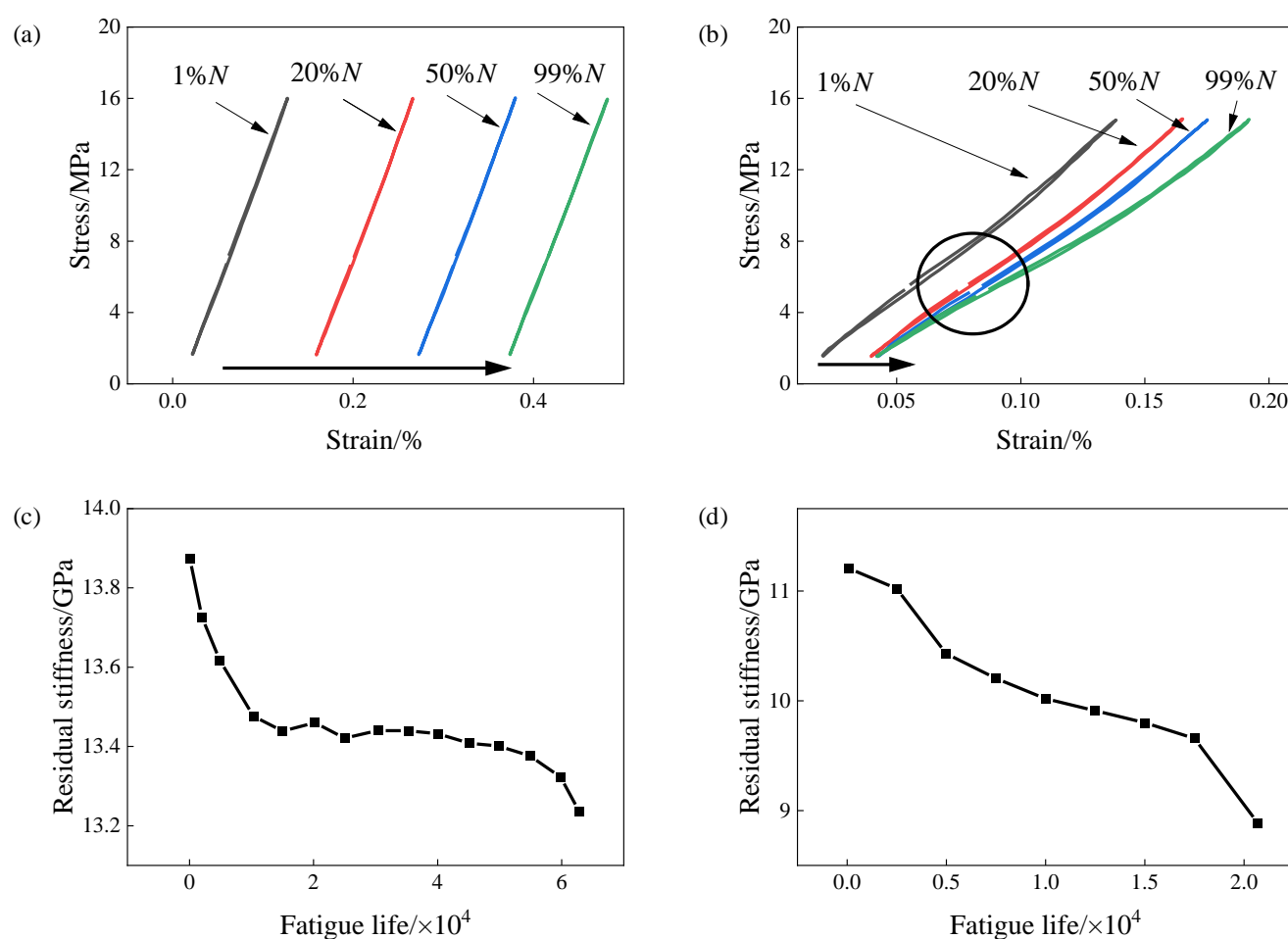


Figure 3. Hysteresis curve evolution and stiffness degradation at different temperatures at 48% stress level. (a) Hysteresis loop at 800°C; (b) Hysteresis loop at 1000°C; (c) Stiffness degradation curve at 800°C; (d) Stiffness degradation curve at 1000°C

Conclusion

Fatigue tests and strength evaluations were conducted on 2.5D SiO_{2f}/SiO₂ at elevated temperatures. Finding reveals that:

- Within the experimental temperature range, both the static strength and the fatigue limit under specified conditions gradually increased.
- The material exhibited increased strength under lower fatigue load due to the fatigue strengthening mechanism. The experimental results from 800°C ~ 1000°C show good agreement with the stiffness degradation model.
- The fracture mechanism of the virgin specimen exhibited brittle fracture, and the pre-specimen demonstrated a combination of brittle and ductile fracture mechanisms.

3. Residual stiffness degradation model

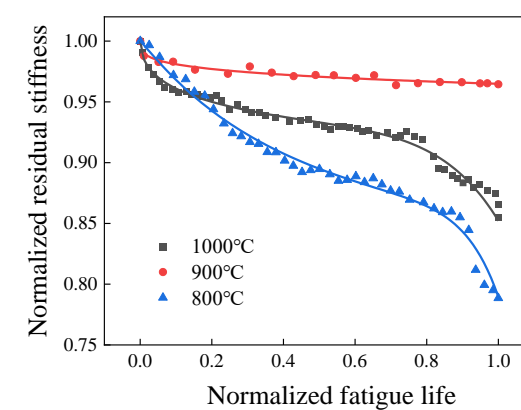


Figure 4. Stiffness test value and model curves of specimens

$$\frac{E_r}{E_0} = 1 - \left(1 - \frac{E_{rc}}{E_0}\right) \left\{1 - \frac{1 - (n/N)^a}{1 + b(n/N)^c}\right\} \quad (1)$$

Table 2. Fitting parameters and goodness-of-fit

T/°C	Stress level	a	b	c	R ²
1000	50%	7.047	1.155	0.5075	0.975
900	45%	0.2598	-0.0102	-0.2723	0.9429
800	48%	12.48	2.086	1.068	0.9858

4. Effect of fatigue load on residual performance

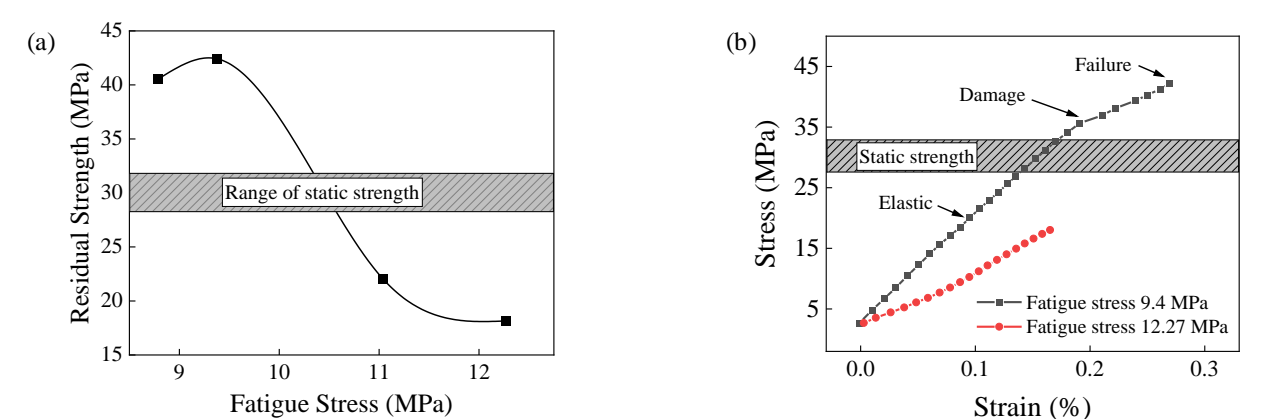


Figure 5. Residual tensile strength result. (a) Residual strength variation curve with fatigue load; (b) Stress-strain curve

5. Damage mechanisms

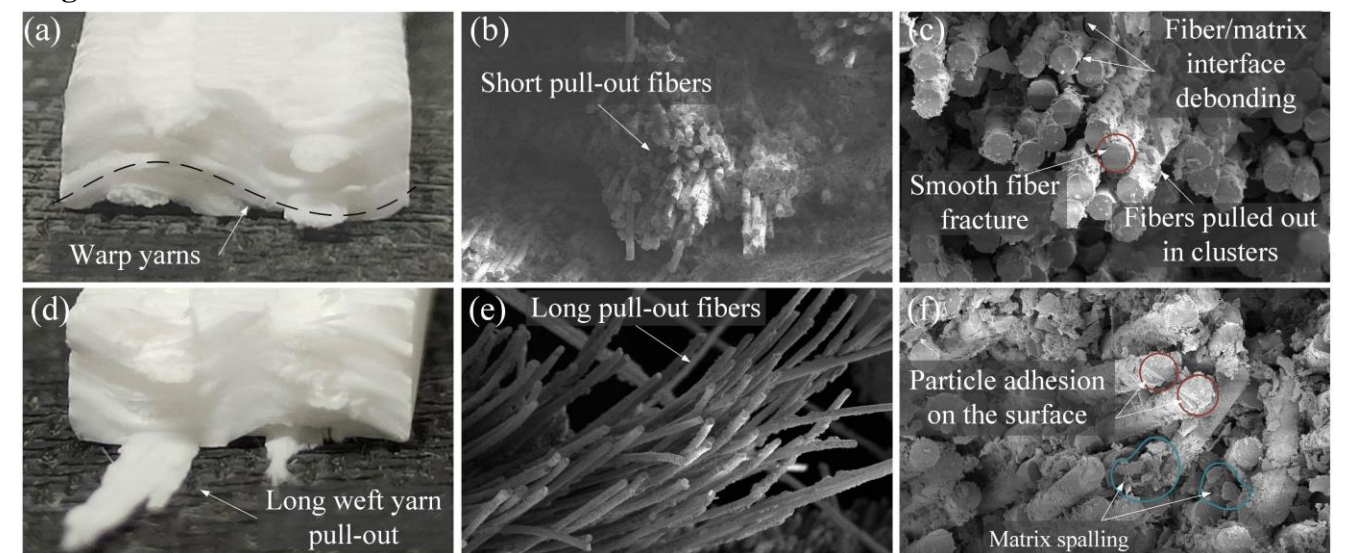


Figure 7. Fracture morphologies. (a) The virgin specimen; (b) The virgin specimen (1000×); (c) The virgin specimen (4000×); (d) The pre-fatigue specimen; (e) The pre-fatigue specimen (1000×); (f) The pre-fatigue specimen (4000×).

Reference

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